

# Study the Increasing of the Cantilever Plate Stiffness by Using Stiffeners

Jawdat Ali Yakoob      Iesam Jondi Hasan

Ass. Prof.                      Ass. Lectu.

Ref. & Air. Cond. Dept/ Technical College / Kirkuk – Iraq-

**Abstract:** The present work is aimed to study the static bending analysis of the rectangular cantilever plate with different shaped stiffeners. The rectangular thin plate is considered and its deflection due to acting load at the free tip is simulated by static analysis using FEM and (ANSYS) software then compared with exact analysis. The analysis is performed for the rectangular plates with different cross-section area stiffeners. The stiffeners area added to the plate area, or the cross section area of the plate with and without stiffeners kept constant to keep the plate weight constant. The plate and stiffeners assumed to be as rigid coupling. The static deflection is performed for different geometries of stiffeners. The boundary conditions for all plates considered as cantilever plate.

In this work, by using of finite element method analysis and exact analysis for thin plate theory, five different shape stiffeners are considered, including square, triangle, trapezoidal and T-section with two different thicknesses. For more easily analysis the plate behavior considered as beam behavior because the plate width is very small in compare with the plate length. For all stiffener types, at first the length, thickness and width of the plate are assumed constant and the area of stiffeners added to the plate section, then in the second case the cross section area of the plate with and without stiffeners assumed as constant, (i.e., the cross section area of the plate and stiffener equal to the flat plate cross section area).

The results showed that the adding of the stiffeners with each geometry shape and size to the plates are increased the plate stiffness. The plate with square stiffener become stiffer than others, but with more added mass and area. The 1mm T-shape stiffener gave stiffness near to the equivalent rectangular plate with lower added mass and area.

**Keywords:** Plate bending, Point load, Stiffeners, Optimum shape.

## Introduction

The reinforced plates are often used in engineering practice because they are stiff, strong, stable and proportionally light. Typical areas of application of such plates are in many fields, such as cars structure, aircrafts, ships, tanks and in many supporting frames. Many industrial elements are made as thin panels reinforced by stiffeners. The stiffeners are different in their geometry, size, cross-section area and mass, and then the choice of an optimal one is important to obtain the required stiffness with keeping the weight of the structure as low as possible.

Adding of stiffener significantly affects the static characteristics of the structure hence it has a direct interest in structure design. Luo and Gea [1], proposed to design the optimal stiffener with three - dimensional trapezoidal shell/plate structure for static and eigenvalue problems. The stiffener location problem is solved by a microstructure-based design domain method and the orientation problem is modeled as an optimization orientation problem of equivalent

orthotropic materials by newly developed energy-based method.

Jarmai & Farkas [2], are studied the minimum cost design of uniaxially compressed plates with welded trapezoidal stiffeners considering the buckling failures. It has been taken the design constraints as global, local and torsional buckling and the limitation of the thickness of cold-formed trapezoidal stiffeners and limitation of the distortion caused by shrinkage of welds. They take the global buckling constraint into account due to the given deviation in different parameters. The optimum dimensions of stiffeners and their numbers are determined by mathematical programming method.

Alinia [3], analyzed a group of 1200 plates by using ANSYS finite element method of analysis to study the role of longitudinal stiffeners to come up with some limits concluded for an optimized design procedure. He is investigated that the number of panels produced by intermediate transverse stiffeners should not be less than the value of plate's aspect ratio. Also the optimum geometric properties of the stiffeners correspond to the point when the

buckling shape of a plate changes from the overall mode to local mode.

Yahya [4], obtained an economical design of plate loaded by pressure using rectangular stiffeners. The effect of stiffeners height on the maximum stress in the plate subjected to pressure load was obtained. To find the effect of stiffeners thickness, plate dimensions, and pressure on the optimum stiffeners height and different sets of plate-stiffeners combinations are selected, using FEM software package ANSYS.

Nildem [5], considered the buckling optimization of stiffened plates under uniform edge compression. The design variables considered as the stiffeners location and its effect on the critical buckling load. Investigated two types of stiffened plates are used in aerospace industry. From the results, the influences of size optimizations and stiffeners location are presented to develop and improve the design for stiffened plates.

Szczepanik and Burczynski [6], are studied the swarm methods and the finite element method to optimize the stiffeners location in the 2-D structure. The criteria's of the optimized structure are stress and displacement. From numerical technique they found that the effective technique for solving the computer aided optimal design is the method based on the swarm computation. They present additional comparisons of the effectiveness of the particle swarm optimizer (PSO) and evolutionary algorithms (EA) methods.

In the present study, the effect of adding single stiffener with deferent geometries to long cantilever plate as rigid coupling is investigated. Finite element method is used to analyze the bending behavior of the plates due to the acted load on free tip. Five solid stiffeners with deferent geometries are considered, including square, triangle, trapezoidal, and two T-section stiffeners (1 and 2 mm) thickness.

### ***Cantilever Plate Bending***

Plate is a flat surface having large dimensions as compared to its thickness. The plate behavior in steady static bending condition approximately likes to the beam behavior when the plate width is very small when compared with the plate length. The bending properties of a plate depend greatly on its thickness. Hence; in classical theory there are thin and thick plates and thin

plate has two sub-theories, small and large deflection theories. In thin plates with small deflections theory, the following assumptions are considered [7]:

- (a) There is no deformation in the mid-surface plane of the plate. This plane remains neutral during bending.
- (b) Points of the plate lying initially normal to the middle surface of the plate remain normal to the same surface even after bending.
- (c) The normal stresses in the transverse direction to the plate are negligible.

This theory is satisfactory for plates with thickness to length ratio exceeding (1/10) and the ratio of maximum deflection to the thickness less than (1/5). Many engineering problems lie under the above conditions and satisfactory results are obtained by classical theories of thin plates.

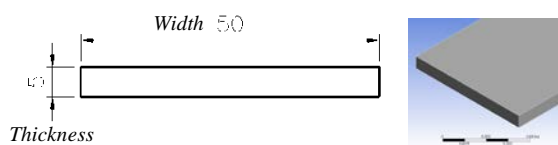
For cantilever plate there is one free end and other clamped end. The cantilever plate will bend laterally when a normal force is acting on the free end tip, this bending depends on plate dimensions, material, and force value. In many engineering examples bending deformation is needed to be reduced. One of the common methods is by reinforcing the plate by using stiffeners instead of increasing plate thickness, and the effect of the stiffeners on the plate bending behavior depends on the dimensions and geometry of the stiffeners, in addition to the method of clamping between the plate and stiffeners.

The using of different geometries of single stiffener to reinforce cantilever plate as a rigid clamping along the plate is included in the present study to determine the optimum stiffeners geometry with low added mass, and analyzing of plate bending value numerically using finite element method then comparing it with exact bending theory.

### ***1- Modeling of Plates***

A thin rectangular plate is considered as shown in figure (1) for the static bending of rectangular plate. The plate is assumed to follow the classical beam bending. The plate's length along x direction, width along y direction and thickness along z direction, hence the plate is assumed to lie in the x-y plane. The plate has two independent displacements (i.e., displacement in y direction and rotation around z axis). The rectangular cross section plates with

different shaped stiffeners are shown in figure(2).



(1000 mm length)

Figure (1): Flat rectangular plate

will be as shown in the figure (3); where the plates thickness are reduced according to the added stiffener cross section area to keep the plate mass constant in all types.

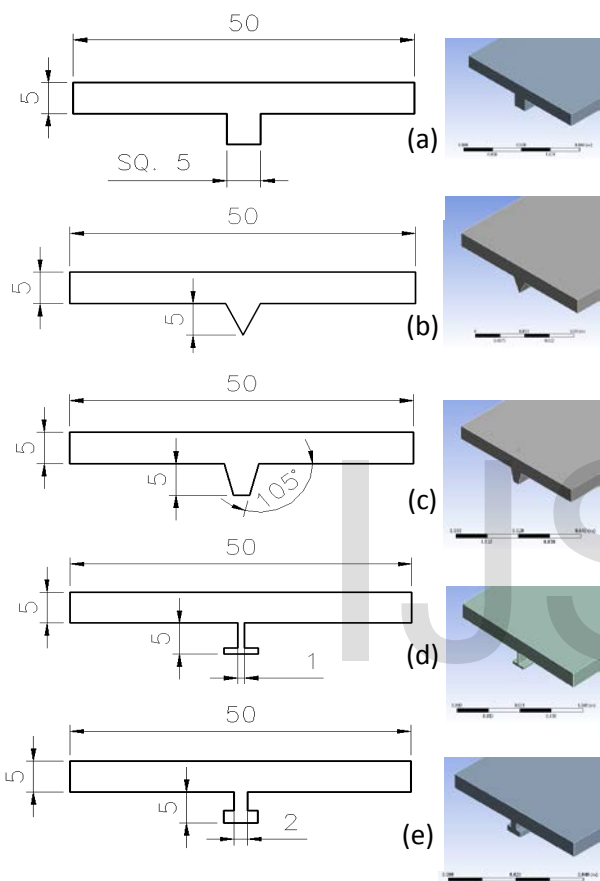


Figure (2): Rectangular Plate with Different Shape Stiffeners. (1000 mm length for all)

The stiffeners are lied on the related plates such that length of the stiffeners is parallel to the length of the plates. The stiffeners and the plates are assumed to have rigid coupling at their connecting interface.

The stiffeners assumed to have the same whole boundaries (i.e., 5mm whole width and 5mm whole thickness). In this case the stiffeners cross section areas are added to the flat plate cross section area, then the plates mass are increased. In the second case, the cross section area of the plate with and without stiffeners is considered constant, then the plates mass are remained constant. Then the plates dimensions

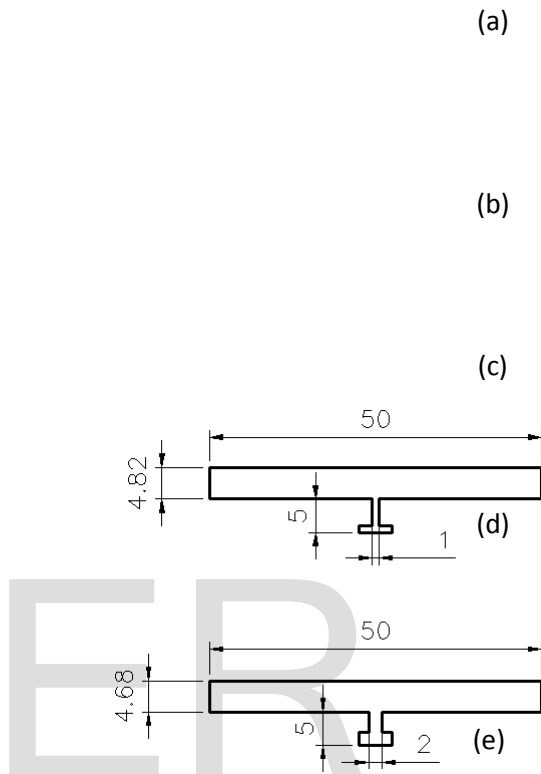


Figure (3): Stiffened Plates with Constant Cross Section.

The plates and stiffeners are assumed to be made of steel with Modulus of Elasticity ( $E = 200$  GPa), Poisson's ratio ( $\nu = 0.3$ ) and Density ( $\rho = 7830$  kg/m<sup>3</sup>).

All plates have the same boundary conditions as cantilever plate (i.e., clamped from one side and free from other side). The plate bending theory is very complex, but whereas the plate width is very small in comparison with the plate length, then the plate can be assumed as a beam, where the more easy beam bending theory can be applied, and in this case the beam bending theory will give accurate results, specially when the plates bend in one plane, (i.e., x-y plane). The plates bends due to lateral force act at their free ends, and then from the strength theory of material, the total deflection ( $\delta$ ) at the free end can be calculated from equation (1) for cantilever beam as [8]:

$$\delta = \frac{W \times L^3}{3 \times E \times I} \quad (1)$$

Where, ( $W$ ) is lateral force at the plate tip, ( $L$ ) is plate length and ( $I$ ) is moment of inertia of the cross section area of the plate. For all plates made of the same material and have the same length, then in equation (1) the lateral force, plate length, and modulus of elasticity are constants, therefore, the formula ( $W \times L^3 / 3 \times E$ ) will be constant, and the moment of inertia changes with the change of the cross section of the plate. Then the moment of inertia ( $I$ ) can be assumed to represent the bending parameter. On the other hand, when the plates made of different materials but have the same length, then the formula ( $W \times L^3 / 3$ ) will become constant, and the formula ( $E \times I$ ) is assumed to represent the bending parameter. Then the equation can be used to calculate the deflection of any plate or beam by changing the bending parameter only.

The moment of inertia of the cross section area of the plate will change with adding the stiffeners to the plate, therefore, the deflection of the stiffened cantilever plate due to lateral force acting at the free end can be calculate by substituting the moment of inertia of the flat plate cross section area with the new moment of inertia including the cross section of the stiffener. The moments of inertia for the two cases of stiffened plates are given in the table (1).

Table (1): Moment of Inertia for All Plates.

No.	Plate with Stiffener type	Moment of Inertia (mm <sup>4</sup> )	
		Case (1)	Case (2)
1	Without Stiffener	520.8333	520.8333
2	Rectangular Stiffener	1141.098	939.427
3	Triangle Stiffener	744.874	657.889
4	Trapezoidal Stiffener	933.152	796.353
5	1mm T-Type Stiffener	841.737	777.971
6	2 mm T-Type Stiffener	1017.388	895.472

## 2- Finite Element Formula

The plates are divided to ( $n$ ) elements, each element has two nodes, and each node has two degrees of freedom including displacement ( $Z$ ) and rotation ( $\theta$ ) as shown in figure (4).

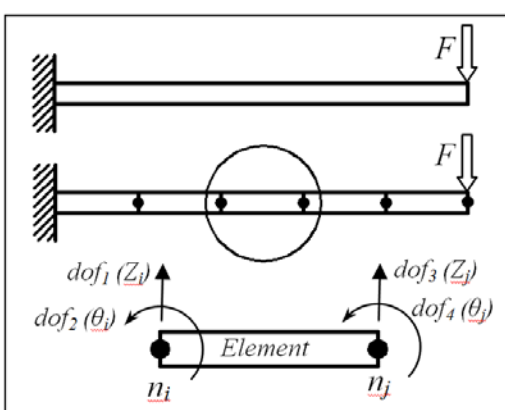


Figure (4): n-element cantilever plate model and node definition

For static deflection the element stiffness matrix can be developed by using basic strength of materials technique to analyze the required force to move each degree of freedom a unit value in the positive direction. Then the stiffness matrix for element ( $i$ ) will take the form [9]:

$$k_{e,i} = E_i I_i \begin{bmatrix} \frac{12}{l_i^3} & \frac{6}{l_i^2} & -\frac{12}{l_i^3} & \frac{6}{l_i^2} \\ \frac{6}{l_i^2} & \frac{4}{l_i} & -\frac{6}{l_i^2} & \frac{2}{l_i} \\ -\frac{12}{l_i^3} & -\frac{6}{l_i^2} & \frac{12}{l_i^3} & -\frac{6}{l_i^2} \\ \frac{6}{l_i^2} & \frac{2}{l_i} & -\frac{6}{l_i^2} & \frac{4}{l_i} \end{bmatrix} \quad (2)$$

Where, ( $k_{e,i}$ ) is element stiffness matrix, ( $E_i$ ) is modulus of elasticity of the element material, ( $I_i$ ) is moment of inertia for the cross section area of the element and ( $l_i$ ) is element length. Then to build the global stiffness matrix ( $k_g$ ) for the plate, the start age will be with ( $n \times n$ ) null matrix, then building up the global stiffness matrix by inserting element after element according to the degree of freedom for each node. In the case of cantilever plate, the degree of freedom of the first node for element (1) is neglected.

## 3- Static Deflection of Cantilever Plate due to Applied Force at the Free End

For static deflection of cantilever plate due to lateral force ( $F$ ) act at the tip of the free end the displacement in ( $Z$ ) direction can be solved. The equation for static equilibrium of the plate is:

$$k_g \times Z = F \quad (3)$$

Where ( $k_g$ ) stiffness matrix, ( $Z$ ) deflections matrix and ( $F$ ) it will be forces matrix. by expanding the equation (3):



$$\begin{bmatrix} k_{g11} & k_{g12} & & & 0 & 0 \\ k_{g21} & 2k_{g22} & \dots & \dots & 0 & 0 \\ & - & & & & \\ & - & & & & \\ 0 & 0 & 2k_{g(n-1)(n-1)} & k_{g(n-1)n} & & \\ 0 & 0 & k_{gn(n-1)} & k_{gnn} & & \end{bmatrix} \begin{bmatrix} Z_1 \\ Z_2 \\ - \\ - \\ Z_{(n-1)} \\ Z_n \end{bmatrix} = \begin{bmatrix} F_1 \\ F_2 \\ - \\ - \\ F_{(n-1)} \\ F_n \end{bmatrix} \quad (4)$$

Where  $(Z_1 \ \& \ Z_2, \dots, Z_{(n-1)} \ \& \ Z_n)$  represents the translation and rotation of nodes with numbers  $(1, 2, \dots, n)$  respectively, and  $(F_1, F_2, \dots, F_{(n-1)}, F_n)$  represents the applied forces on each node. To find each node displacement the inverse of the global stiffness matrix  $(k_g)$  must be calculated, and equation (3) will become:

$$Z = (k_g)^{-1} \times F \quad (5)$$

In the case of cantilever plate with lateral applied force at the free end, all forces value for the forces matrix is equal to zero except  $(F_n)$  which equal to the applied force at the free end.

The solving of the equation (5) performed by using Matlab program. Also the finite element models of all plates are generated by ANSYS software using the Solid45 element. This element is used for generating and meshing solid body; it has four sides, eight nodes and three degree of freedom for each node. The static bending analysis is performed for all cantilever plates.

### Results and Discussion

The static bending analysis is performed for the flat plate and stiffened plates which have one clamped end and the other free end by applying lateral force at the free ends. The lateral applied forces are (10, 20, 30, 40 and 50 N) respectively. The results of five deformation for all six plates types including case (1) and (2) are compared as shown in the tables (2) and (3). It is observed that with addition of stiffener of any type or size tend to increase in the stiffness of the plate in such away more than that of the plate without stiffener.

The effective parameter for cantilever plate bending is the moment of inertia (I), where the

cantilever plate deflection due to the lateral forces acting at the free end are increased with decreasing the moment of inertia as shown in figure (5) for case (1), and figure (6) for case (2). The increase of the deflection average with increasing the force value for the stiffed plate which has less moment of inertia value was more than the plate has more moment of inertia value.

The comparison of bending deformation of the cantilever plates for case (1) is shown in the figure (7), it can be observed that the plate with rectangular stiffener showed lowest deflection with highest moment of inertia, but the plate with triangular stiffener showed highest deflection with lowest moment of inertia. On the other hand, the square stiffener is associated with highest equivalent added mass, but the triangular stiffener is associated with lowest equivalent added mass. For stiffened plates, the square stiffener associated with highest added mass, but the 1mm-T stiffener associated with lowest added mass. Table (4) shows the equivalent flat plate thickness, the added area and added mass caused by using each stiffener type. Where the 1mm-T stiffener has the lowest stiffener added area and lowest ratio between the stiffeners to equivalent added area for equivalent plate.

In the case (2) where the cross-section area and mass of all plates remained constant with and without using stiffeners, the deflection for all plates is shown in the figure (8). In this case the increasing of the plate stiffness was less than those belong to case (1), but without adding additional mass to the plate. Also as in case (1) the stiffness of the plate became larger than un-stiffened plate.

Table (2): Stiffened cantilever plate deflection for case (1).

Force (N)	Plates with Stiffeners Type	Plate without stiffener	With Square Stiffener	With Triangular Stiffener	With Trapezoidal Stiffener	With 1mm-T Stiffener	With 2 mm-T Stiffener
10	Exact	0.032	0.0146058	0.0223755	0.0178606	0.0198003	0.0163818

	FEM	0.032	0.0146	0.0224	0.0179	0.0198	0.0164
	ANSYS	0.031778	0.014553	0.022253	0.017786	0.019712	0.017366
20	Exact	0.064	0.0292116	0.0447503	0.0357212	0.3960065	0.0327636
	FEM	0.064	0.0292	0.0448	0.0357	0.0396	0.0328
	ANSYS	0.063557	0.029105	0.044506	0.035571	0.039423	0.034733
	Exact	0.096	0.0438175	0.0671255	0.0535818	0.0594010	0.0491455
30	FEM	0.096	0.0438	0.0671	0.0536	0.0594	0.0491
	ANSYS	0.095335	0.043658	0.066759	0.053357	0.059135	0.052099
	Exact	0.128	0.0584233	0.0895006	0.0714425	0.0792013	0.0655273
	FEM	0.128	0.0584	0.0895	0.0714	0.0792	0.0655
40	ANSYS	0.12711	0.058211	0.089013	0.071143	0.078846	0.069466
	Exact	0.16	0.0730291	0.1118758	0.0893031	0.0990016	0.081909
50	FEM	0.16	0.0730	0.1119	0.0893	0.0990	0.0819
	ANSYS	0.15889	0.072763	0.11127	0.088928	0.098558	0.086832
Rip Area (mm <sup>2</sup> )		-	25	12.5	18.3	9	16

Table (3): Equivalent plate thickness and added stiffeners area for case (1).

Stiffener Type	Equivalent Thick. (mm)	Added Thick. (mm)	Equivalent Added area (mm <sup>2</sup> )	Added Stiffener area (mm <sup>2</sup> )	Stiffener /Equivalent (%)	Added Stiffener mass (%)
1 Flat	5	0	0	0	0	0
2 Square	6.493986757	1.493986757	74.69933785	25	33.4674988	10
3 Triangle	5.633323497	0.633323497	31.66617485	12.5	39.47429729	5
4 Trapezoidal	6.072784712	1.072784712	53.6392356	18.3	34.11681728	7.32
5 2mm T	6.250277353	1.250277353	62.51386765	16	25.59432107	6.4
6 1mm T	5.867278	0.867278	43.3639	9	20.7545908	3.6

Table (4): Stiffened cantilever plate deflection for case (2).

Force (N)	Plates with Stiffeners Type	Plate without Stiffener	With Square Stiffener	With Triangular Stiffener	With Trapezoidal Stiffener	With 1mm-T Stiffener	With 2 mm-T Stiffener
10	Exact	0.032	0.01774	0.02533	0.02093	0.02142	0.01861
	FEM	0.032	0.0177	0.0253	0.0209	0.0214	0.0186
	ANSYS	0.031778	0.017682	0.025197	0.020882	0.021329	0.018653
20	Exact	0.064	0.03548	0.05067	0.04186	0.04285	0.03723
	FEM	0.064	0.0355	0.0507	0.0419	0.0428	0.0372
	ANSYS	0.063557	0.035363	0.050394	0.041764	0.042658	0.037306
30	Exact	0.096	0.05322	0.0760	0.06278	0.06427	0.05584
	FEM	0.096	0.0531	0.0760	0.0628	0.0643	0.0558
	ANSYS	0.095335	0.053045	0.075591	0.062645	0.063986	0.055959
40	Exact	0.128	0.07097	0.1013	0.08372	0.08569	0.07445
	FEM	0.128	0.0710	0.1013	0.0837	0.0857	0.0744
	ANSYS	0.12711	0.070727	0.10079	0.083527	0.085315	0.074611
50	Exact	0.16	0.08871	0.1267	0.10464	0.10712	0.09306
	FEM	0.16	0.0887	0.1267	0.1045	0.1071	0.0931
	ANSYS	0.15889	0.088408	0.12599	0.10441	0.10664	0.093264
Rip Area (mm <sup>2</sup> )		-	25	12.5	18.3	9	16

Table (5): Equivalent plate thickness and stiffener areas for case (2).

Type	Equivalent Thick. (mm)	Added Thick. (mm)	Equivalent Added area (mm <sup>2</sup> )	Stiffener area (mm <sup>2</sup> )	Stiffener /Equivalent (%)	Stiffener mass (%)
1 Flat	5	0	0	0	0	0
2 Square	6.0865	1.0865	54.328	25	46.01678692	10
3 Triangle	5.40516	0.405156	20.2578	12.5	61.70462735	5
4 Trapezoidal	5.7601	0.7601	38.005	18.3	48.151559	7.32
5 2mm T	5.9902	0.9902	49.5076	16	32.31827033	6.4
6 1mm T	5.7158	0.7158	35.792	9	25.14528386	3.6

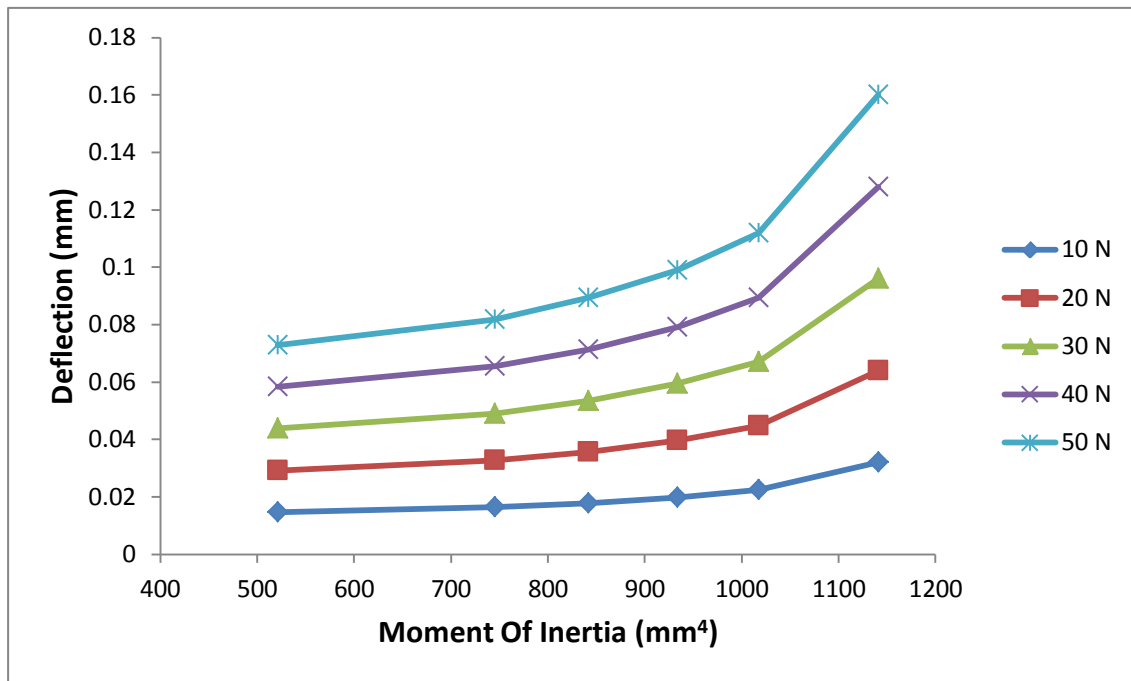


Figure (5): Effect of moment of inertia on the plate deflection for case (1).

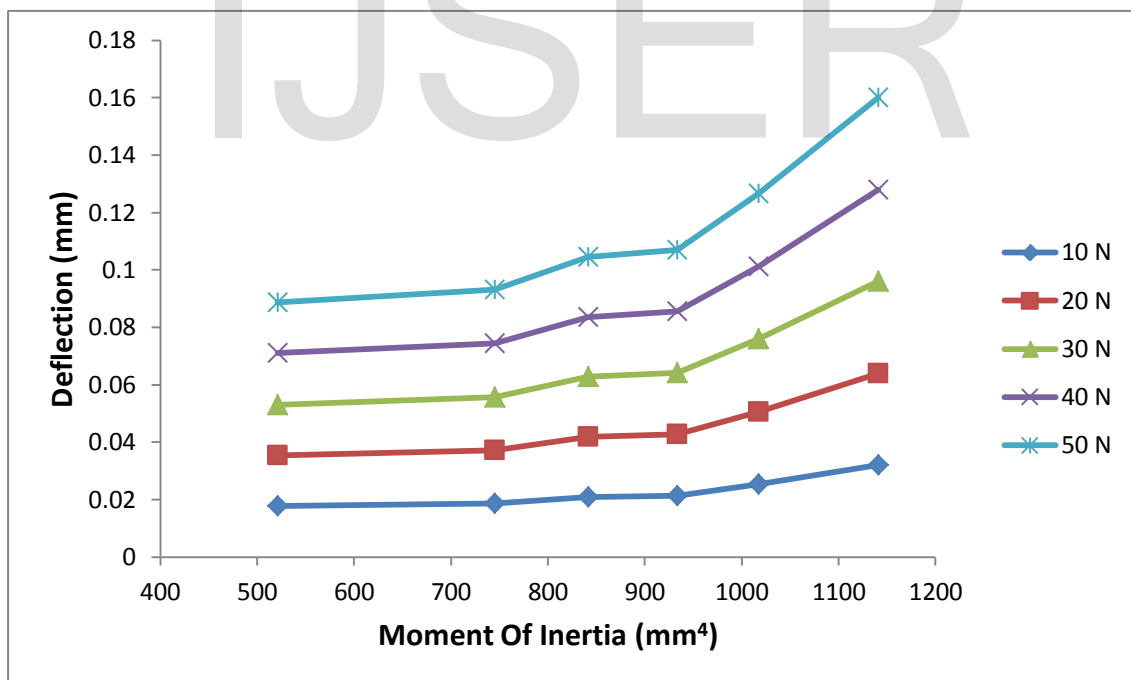


Figure (6): Effect of moment of inertia on the plate deflection for case (2).



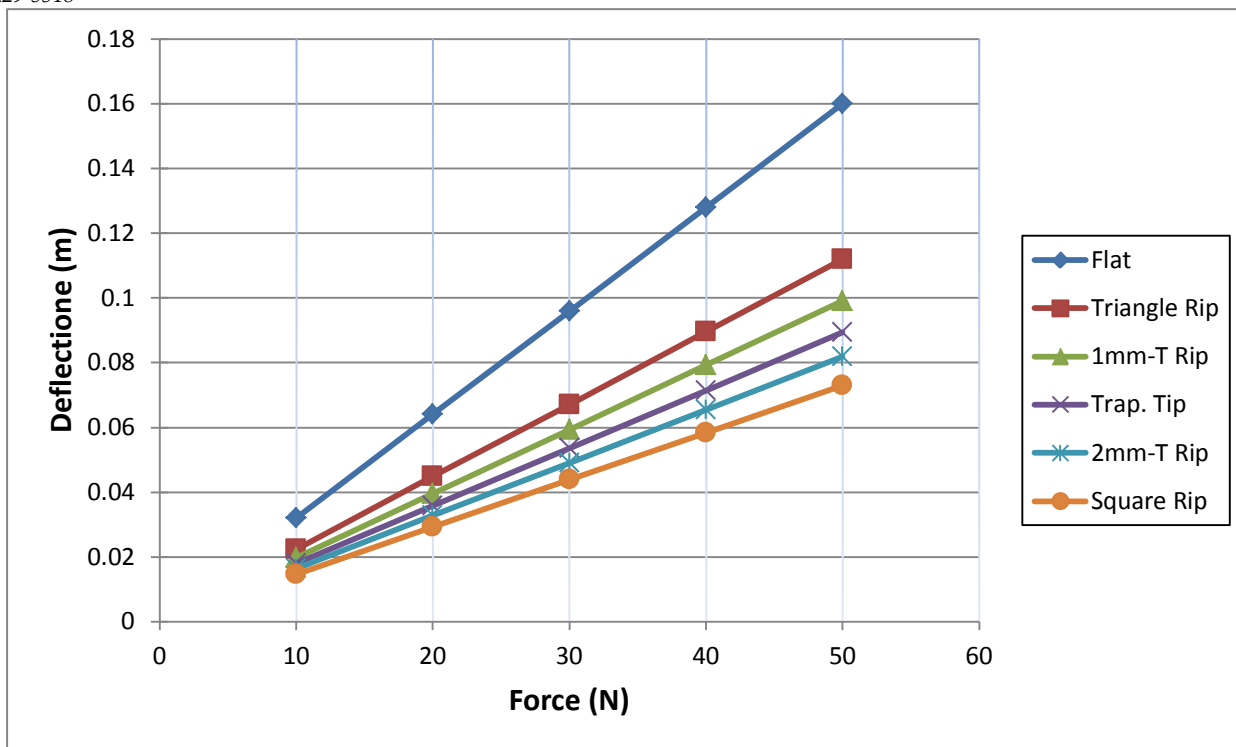


Figure (7): Plates Deflection for Case (1) for Each Stiffener Type.

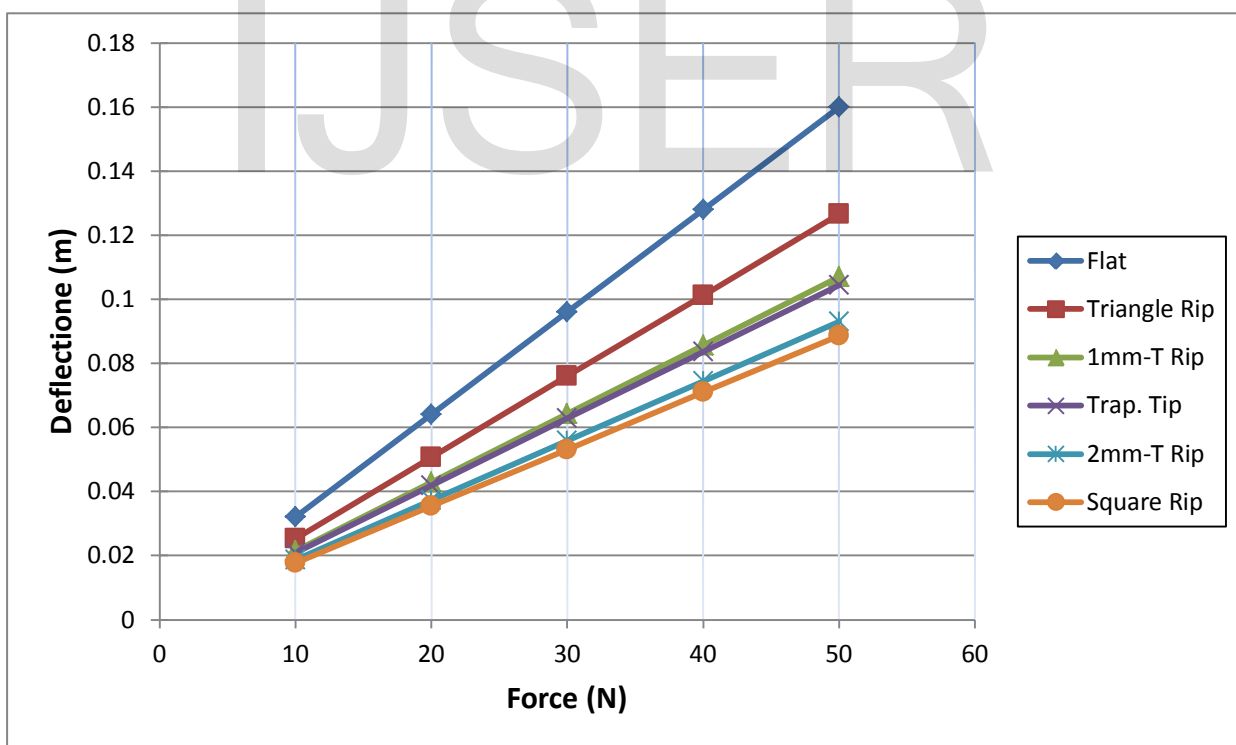


Figure (8): Plates Deflection for Case (2).

**Conclusion**

The static bending analysis for a plate without stiffener and stiffened plates is performed.

Results of the analysis gave the following information regarding static bending of plates:

- 1- Addition of a stiffener to the plate with any shape and size has an effect of increasing the

- plate stiffness to resist the bending due to lateral force.
- 2- The moment of inertia is the effective parameter on the plate bending (bending parameter) for the plates made from the same material and with same dimensions.
  - 3- The case of adding stiffener to the plate gave highest stiffness increasing in comparison with the case of adding stiffener with remaining the plate mass constant with and without stiffener.
  - 4- The using of square stiffener caused the highest increasing in the stiffness value but with highest added mass.
  - 5- The 1mm-T type stiffener gave suitable stiffness to the plate with lowest added mass in comparison with other stiffener types.
  - 6- The triangular stiffener type has the lowest equivalent plate thickness.

- 7- Timoshenko, S. and S. Woinowskky-Krieger. "*Theory of Plates and Shells*". McGraw Hill. pp. 21-46.
- 8- Khurmi, R. et al. "*Theory of Machines*", 14<sup>th</sup> edition, S. Chand & Co. Ltd., New Dehli 2005;. pp. 919.
- 9- Saeed Moaveni. "*Finite Element Analysis. Theory and Application with ANSYS*". Prentice. Hall., Inc, 1999. pp.328-329.

## References

- 1- J. Luo and H.C. Gea. "*A Systematic Topology Optimization Approach for Optimal Stiffener Design*". Structural Optimization 16, © Springer – Velarg, 1998. pp. 280-288.
- 2- K. Jarmai & J. Farkas. "*Minimum Cost Design of Uniaxially Compressed Plates with Welded Trapezoidal Stiffeners Considering a Reliability Constant*". Metal Structures – Design, Fabrication, Economy, Jarmai & Farkas (eds) © 2003 Millpress, Rotterdam, ISBN 90-77017-75-5. pp. 209-210.
- 3- M.M. Alinia. "*A Study Into Optimization of Stiffeners in Plates Subjected to Shear Loading*". Elsevier Ltd. 2005. pp. 845-846.
- 4- Yahya Mohey Hamad. "*Optimum Height of Plate Stiffener Under Pressure Effect*". The 1<sup>st</sup> Regional Conference of Eng. Sci. NUCEJ Spatial ISSUE Al-Nahrain University/ College of Engineering. Voll. 11, No. 3. 2008. pp. 459-460.
- 5- Nildem Taysi. "*Determination of Thickness and Stiffener Location for Optimization of Critical Buckling Load of Stiffened Plates*". Scientific Research and Essays Vol. 5(9), 4 May 2010. pp. 897-910.
- 6- M. Szczepanik and T. Burczynski. "*Swarm Optimization of Stiffeners Locations in 2-D Structures*". Bulletin of the Polish Academy of Sciences Technical Sciences, Vol. 60, No. 2,2012. Pp. 241.